Rascal Requirements Verification Document

Saint Louis University

Rascal



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Copper Operational

Test Plan

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# INTRODUCTION

The Rascal Requirements Verification Document serves to define and elaborate on each of the requirements laid out in the RCL-RVM-CMQA1 Rascal Requirements Verification Matrix (RVM), as well as the ways in which each of said requirements will be validated prior to and during the Rascal mission.

The Rascal Mission, as defined by the RVM, can be broken down into Six Stages, with each stage having a set of requirements directly associated with it. Within the requirements associated with each stage are sub-requirements associated with a particular subsystem of the Rascal CubeSat system. The manner in which each of these requirements is associated with a particular stage or subsystem is through a simple notation scheme, as discussed below:

|  |  |  |  |
| --- | --- | --- | --- |
| Mission Identifier | Stage Identifier | Sub-System Identifier | Requirement Number |
| RCL |  |  |  |

# NOMENCLATURE

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# REQUIREMENT STAGES

# Pre-Launch Requirements

1. **The Total CubeSat System Volume Shall Not Exceed 6U**
2. **The Total CubeSat System Mass Shall Not Exceed 8.0 kg**
3. **All Materials Used in the CubeSat System shall have a Total Mass Loss of Less Than 1.0%**
4. **All Materials Used in the CubeSat System Shall Have a Collected Volatile Condensable Material of less than 0.1%**
5. **The CubeSat System Must be in Orbit for at Least 6 Months**
6. **The CubeSat System Must Deorbit within 25 Years of being Launched**
7. **Jade and Ruby Shall be Conjoined Prior to Launch Vehicle Integration**
8. **The CubeSat System Shall Incorporate a Deployment Switch**
9. **No Protrusion Shall Extend beyond 6.5 mm Normal to Any External Surface of Jade or Ruby**
10. **No External Components Other than the CubeSat Rails of Jade and Ruby may make Contact with the Deployer**
11. **The Deployer Shall not be Used to Secure Any CubeSat Deployables**
12. **The Center of Gravity of the total CubeSat System Shall be Located within a Sphere of 2 cm of the Geometric Center of the System**
13. **The Center of Gravity of Jade and Ruby Shall be Located within a Sphere of 2 cm of their Geometric Center**
14. **The CubeSat System Coordinate System Shall be Defined As Specified in Figure 1-1**
15. **The Local Coordinate System of Jade and Ruby Shall be Defined as Specified in Figure 1-1**
16. **The Ends of the Rails on the +Z/-Z Faces of the CubeSat System Shall have a Minimum Surface Area of 6.5 mm x 6.5 mm**
17. **The +Y/-Y Faces of Ruby and Jade Shall have a Length of 100 mm**
18. **Jade and Ruby Shall be Capable of Determining Relative Displacement between Each Other**
19. **The CubeSat System Shall be Capable of Recording Relative Displacement Data between Jade and Ruby**
20. **Low Friction, 2D Testing of the CubeSat System Release Mechanism Shall be Conducted**
21. **All Pressure Vessels Shall have a Factor of Safety of No Less Than 4**
22. **All CubeSat Components Shall be Rated to Operate within Temperature Range of at least -20⁰C to 70 ⁰C**
23. **Static Thrust Testing Shall be Performed with the Flight Version of All Pressure Vessels at a Pressure No Greater than 1x10-4 Torr Prior to CubeSat Integration**
24. **All Pressure Vessels Must Pass Thermal Cycle Testing between Temperatures of -30 ⁰C and 70 ⁰C for at Least Two Cycles or for 10 Hours**
25. **Low Friction, 2-D Dynamic Thrust Testing Shall be Conducted with All Pressure Vessels**
26. **The CubeSat System Must Survive Random Vibration Testing Relative to the NASA GEVS Qualification Profile**
27. **The CubeSat System Shall be Subjected to a Temperature of 60 ⁰C at a Pressure No Greater than 1x10-4 Torr for a Minimum of 6 Hours**
28. **The CubeSat System Shall be Able to Execute All Commands Associated with its Operation over RF**
29. **The CubeSat System Shall be Able to Close a Link with the SSRL Ground Station from a Distance of at least 200 meters**
30. **The CubeSat System Shall be able to Document the Functionality of Each of its Subsystems through the Running of a Full-Functional Test**

# Post-Launch Ejection Requirements

1. **The CubeSat System Shall not Broadcast in RF Until Ejection +45 Minutes**
2. **The CubeSat System Shall not Release Deployables Until Ejection +45 Minutes**
3. **The CubeSat System Shall Establish Communication Between Itself and the SSRL Ground Station**
4. **The CubeSat System Shall Pass a Health Check Administered from the SSRL Ground Station**

# Separation and Stabilization Requirements

1. **Jade and Ruby Shall be Capable of Separating from One Another with a Relative Velocity of No Greater than 5 cm/s**
2. **Jade and Ruby Shall Achieve a Local Slew Rate of Less than 1 deg/s**
3. **Jade and Ruby Shall Continuously Record Relative Displacement Data Between Each Other**

# Stationkeeping Requirements

1. **Jade and Ruby Shall be Able to Stationkeep within a 10-75 meter Sphere of Each Other for at Least 5 Orbits**

# “Escape” Requirements

1. **Jade and Ruby Shall be Able to Perform an “Escape” Maneuver that Increases the Relative Displacement Between Each Other to at Least 100 Meters within 1 Orbit**

# Rendezvous Requirements

1. **Jade and Ruby Shall be Able to Perform a Rendezvous by Decreasing the Relative Displacement Between Each Other to Within 50 meters for at Least 5 Orbits**